			Certifi	Certified Repair Station	
			Phone	(	
			Fax	(	
	<u>-</u>	Cycle Sheet			
W.O.			Date		
•	A/C Model	Eng Model	Eng S/N (PCE#)	Tach/Hobbs	
	N/A	PT6A-28		N/A	
Eng TT	Eng TSO	Eng CSN	Eng CSO	Eng TSHSI	
16,557.3	2,193.3	15,965	1,830	0.0	

## Work Performed

- Sudden Stoppage Inspection
- Hot Section Inspection
- Clean and Flow Fuel Nozzles
- Test Cell

Parts List Attached

Remarks: The following installed new: 41 PT Blades P/N E3013102, 16 Shroud Segments P/N PF3019438CL9, and No. 3 Bearing Cover P/N PTC3104278-01 S/N PTC0816. The following installed in overhauled condition: No. 6 Bearing P/N 3007147 S/N R3299, CT Vane Ring P/N 3032651CL13.1 S/N S1-580004, CT Disk P/N 3013411 S/N 4B083, Exhaust Duct P/N 3101037-02 S/N 5E095, Prop Governor P/N 8210-002-01 S/N 16083454, and PT Vane Ring P/N 3024862CL12.7.

			Cyçles	
Nomenclature	P/N	S/N		
			Consumed	Remaining
1 <sup>st</sup> Stg. Hub	3013111		15,965	3,035
2 <sup>nd</sup> Stg. Disk	3013712		15,965	8,035
3 <sup>rd</sup> Stg. Disk	3011713		15,965	9,035
Impeller	3013176		15,965	3,035
CT Disk	3013411		13,499	2,511
1st Stg. PT Disk	3022412		15,965	4,034

All work carried out I.A.W. P&WC

Maintenance Manual P/N 3013242 Rev. 46

Overhaul Manual P/N 3013243 Rev. 40

This signed document is an official Log Book Entry and must be fixed in the appropriate Log Book

This certifies that only work approved by the customer was accomplished I.A.W. FAA regulations and current Mg. Specifications. The product is approved for return to service. Pertinent details are on file at this repair station under the assigned Work Order (W.O.) number (if applicable). Turbine Blade Warranty: New blades are warranted by manufacturer; overhauled blades are warranted by overhaul vendor; serviceable blades are inspected with populated PAWC Manuals and are reinstalled as per customer.

**Authorized Signature** 

Mendozé/ Chief Inspector

ite <u>6/2</u>

6/2**8**/18

Á	proving Civil Aviation uthority/Country: UNITED STATES	<sup>2.</sup> AU	THORIZED REL FAA FORM 8130-3, AIRWO	3. Form Tracking Number:		
4. Organizat	ion Name and Address:					5. Work Order, Contract, or Invoice Number:
6. Item:	7. Description:		8. Part Number:	9. Quantity:	10. Serial/Batch Number:	11. Status/Work:
1	Engine Asse	mbly	3015000	1		Repaired
TSN: 16,557.3 TSO: 2,193.3 CSN: 15,965 CSO: 1,830  Sudden Stoppage Inspection Hot Section Inspection Clean and Flow Fuel Nozzles Test Cell  The following installed new: 41 PT Blades P/N E3013102, 16 Shroud Segments P/N PF3019438CL9, and No. 3 Bearing Cover P/N PTC3104278-01 S/N PTC0816. The following installed condition: No. 6 Bearing P/N 3007147 S/N R3299, CT Vane Ring P/N 3032651CL13.1 S/N S1-580004, CT Disk P/N 3013411 S/N 48083, Exhaust Duct P/N 3101037-02 S/N 5E095, Prop Governor P/N 8210-002-01 S/N 16083454, and PT Vane Ring P/N 3024862CL12.7.  All work performed in accordance with Pratt and Whitney Canada Maintenance Manual P/N 3013242R46 and Overhaul Manual P/N 3013243R40. Pertinent details are on file at this repair station under the assigned work order.						
□ Ap □ No	proved design data and are in on-approved design data spec rized Signature	TQZR133K				
13d. Name	(Typed or Printed);	13e. Date	e (dd/mmm/yyyy):	14d. Name <i>(Type</i> Je:	d or Priotid) sus Mendoza	14e. Date (dd/mmm/yyyy): 28 Jun 2018

FAA Form 8130-3 (02/14)

User/Installer Responsibilities

It is important to understand the existence of this document alone does not automatically constitute authority to install the aircraft engine/propeller/article.

Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts aircraft engine(s)/propeller(s)/articles(S) from the airworthiness authority of the country specified in Block 1.

Statements in Blocks 13a and 14a do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the nethods. regulations by the "ser/installer before the aircraft may be flown.

**Customer:** 

**Job Number:** 

Engine Model: PT6A-28

**Serial Number:** 

Date: 06/28/2018

Rating	S.H.P	NG Speed	S.F.C	TT5/TH °R		
Performance Ref. Data	680	36550	.597	1615		
Data Plate T5 Trim: 71 °F 39.4 °C OHMS: 14.0  Trim Stick: 3031417 CL 40						
CT Stator Flow Class: 13.10						
PT Stator Flow Class: 15.44						

Oil Type Used: BP2380

Fuel Type Used: JET A

I hereby certify that this engine has been tested in accordance with the manufacturer's recommendations and requirements.

Signature: \_

Tate: